

Analytical Study on Space Debris and its Impact the Growing Threat of Space Debris to Sustainable Space Exploration

Sushil S. T.¹; Santhanalakshmi V.²; Dr. D. P. Sivasakti Balan³;
R. J. Thayumanaswamy⁴

¹(Grade 11), ²(Mentor), ³(Principal), ⁴(CEO)

^{1,2,3,4}KRM Public School, Sembium

Publication Date: 2026/09/18

Abstract: The rapid expansion of space activities over the past decades has led to a significant increase in the population of artificial objects in Earth's orbit, commonly referred to as space debris. This research paper presents a comprehensive analytical study of space debris, focusing on its origin, classification, distribution, and dynamic behaviour in various orbital regimes, including Low Earth Orbit (LEO), Medium Earth Orbit (MEO), and Geostationary Orbit (GEO). The study synthesizes data from international space agencies and tracking systems to evaluate debris density, size distribution, and collision probabilities. A key aspect of this research involves modelling the evolution of debris populations using statistical techniques. The paper examines major contributors to debris generation, such as satellite fragmentation events, defunct spacecraft, launch vehicle remnants, and collisions. Special attention is given to cascading collision phenomena (Kessler Syndrome), which poses a long-term threat to the sustainability of space operations. The impact analysis explores both operational and environmental consequences. For active satellites and space missions, debris presents severe risks including structural damage, mission failure, and increased operational costs due to collision avoidance manoeuvres. The study also evaluates risks to human spaceflight and the integrity of critical infrastructure such as communication, navigation, and Earth observation systems. Furthermore, the paper discusses potential re-entry hazards and their implications for Earth's surface. Mitigation and remediation strategies are critically assessed, including international guidelines, debris minimization techniques, end-of-life disposal practices, and emerging technologies for active debris removal (ADR). Policy frameworks, leading countries that contribute to space debris and global cooperation efforts are also analysed to highlight the importance of sustainable space governance. The findings emphasize the urgent need for coordinated international action, improved tracking capabilities, and the adoption of robust mitigation strategies to ensure the long-term usability of outer space.

Keywords: Low Earth Orbit (LEO), Medium Earth Orbit (MEO), and Geostationary Orbit (GEO), Kessler Syndrome, Active Debris Removal (ADR).

How to Cite: Sushil S. T.; Santhanalakshmi V.; Dr. D. P. Sivasakti Balan; R. J. Thayumanaswamy (2026) Analytical Study on Space Debris and its Impact the Growing Threat of Space Debris to Sustainable Space Exploration. *International Journal of Innovative Science and Research Technology*, 11(9), 359-365. <https://doi.org/10.38124/ijisrt/26sep040>

I. INTRODUCTION

Space debris poses operational risk to aerospace missions, particularly in low earth orbit. Agencies like NASA, ESA, SPACE X, and JAXA are concerned about the increased rate of junks in space orbiting our earth.



Fig 1 Man-Made Objects that are Left in the Space.

Space debris often referred to as space junk encompasses all human-made objects that remain in Earth's orbit but no longer serve any functional role.

This category is remarkably diverse, including large items such as defunct satellites, abandoned spacecraft, and spent rocket stages, as well as much smaller fragments like bolts, shards of metal, and even microscopic paint flecks.

Over decades of space exploration, these remnants have accumulated, creating a growing cloud of clutter around our planet. While some pieces are massive and easily trackable, others are so tiny that they are nearly invisible yet still capable of causing damage due to the high speeds at which they travel. Collectively, this debris poses risks to operational satellites, space stations, and future missions, making it a significant challenge for space agencies and researchers working to ensure safe and sustainable use of Earth's orbital environment.

- Low earth orbit (LEO): 300 to 2,000 km altitude, 7 to 8 km/s orbital speed, 1.5 - 3 hour period.
- Semi-synchronous: 20,000 km altitude, 4 km/s orbital speed, Navigation satellites, 12 hour period.
- Geosynchronous: 36,000 km altitude, 3 km/s orbital speed, Communication or broadcast satellites, 24 hour period.

II. REVIEW OF LITERATURE

The literature review contains decades of observations, theory, and recent experiments to explain how the orbital environment has evolved, why certain regions are unstable, and which responses are most promising. It summarizes the historical buildup of cataloged and untracked debris and the role of fragmentation events in driving collision risk, then outlines Kessler-type cascade dynamics that make prevention and targeted removal more effective than purely reactive measures. The review also covers advances in space-situational awareness radar, optical, laser ranging, and in-situ sensors and evaluates active removal concepts (drag augmentation, tethers, nets/harpoons, ion-beam shepherds, and lasers), comparing their operational envelopes, risks, and scalability. Finally, it surveys probabilistic and data-driven methods (Monte Carlo ensembles and machine-learning analyses) that identify the orbital and design parameters most critical to collision probability and satellite lifetime.

➤ *Historical Growth and Inventories*

Long-term analyses show that the population of human-made objects in Earth orbit has grown steadily since Sputnik's launch in 1957, punctuated by sudden surges following fragmentation events and anti-satellite tests. While the catalog of trackable, meter-class objects now numbers in the tens of thousands, modeling and in-situ measurements indicate a far larger cloud of smaller debris ranging from millimetres to centimetres that is not routinely tracked and therefore remains a hidden hazard.

These small fragments are extremely numerous and travel at orbital velocities of several kilometres per second, so even a particle only a few millimetres across can deliver destructive kinetic energy on impact. In particular, medium-sized fragments (roughly 5 mm–10 cm) present a disproportionate threat: they are often too small to be reliably tracked from the ground yet large enough to penetrate shielding or disable critical systems, making them a primary driver of collision risk for operational spacecraft. This size-dependent visibility and lethality combined with the concentration of objects in popular altitude bands like LEO and GEO means that untracked medium fragments dominate encounter flux and complicate both avoidance maneuvers and long-term mitigation planning.



Fig 2 Increasing Rate of Space Debris.

The figure above shows the significant and profound increase in the number of space debris which are accumulated on the orbit of the earth. Presently millions of space junk orbiting the Earth at speeds up to several kilometers per second.

➤ *Risk Theory and Kessler Syndrome*

The theoretical foundation for understanding long-term debris growth was established by Kessler and Cour-Palais in 1978, who proposed that, beyond a critical density of objects, collisions between debris and satellites would generate fragments faster than natural atmospheric drag and other processes could remove them. This self-sustaining, cascading process now known as the Kessler Syndrome implies that even a complete halt in future launches would not by itself stop debris growth in the most congested orbital bands, since collisions among existing objects would continue to seed new fragments. Debris is generally classified by size into three practical categories: small (below roughly 5 mm), which is non-trackable but can often be mitigated with shielding; medium (roughly 5 mm to 10 cm), which is non-trackable from the ground yet energetic enough to disable or destroy a spacecraft on impact; and large (above 10 cm), which is trackable and can usually be avoided through planned manoeuvres. Because collisions between large, trackable objects are comparatively rare occurring on the order of once every five to eight years by some estimates but generate very large numbers of new fragments each time, such events disproportionately drive the long-term growth of the medium and small debris populations that are hardest to observe and avoid. Research consistently finds that compliance with debris-mitigation guidelines needs to remain above roughly

90 percent across the international space community for the orbital environment to be stabilised, and current agency reporting confirms this threshold has still not been reliably met at a global scale.

➤ *Space Situational Awareness and Measurement Advances*

A growing body of national and agency reporting documents steady improvement in the tools used to detect, track, and characterise orbital debris. JAXA, for example, operates a dedicated space situational awareness (SSA) system combining a purpose-built low-Earth-orbit radar capable of detecting 10-cm-class objects at 650 km altitude, refurbished 1-metre and 50-centimetre optical telescopes for higher-orbit and geostationary observation, and a satellite laser ranging (SLR) station at Tsukuba alongside an in-situ micro-debris sensor (the Space Debris Monitor) that can register impacts from particles as small as 100 micrometres. Italy's national space agency has similarly deepened its observational capacity through the ASI-INAF and ASI-PoliMI research agreements, which combine a national network of optical telescopes, a dedicated robotic camera array (the ASTRA project) for fast-moving low-orbit objects, enhanced radar tracking, and artificial-intelligence tools for real-time detection and orbital prediction.

Slovakia's Comenius University operates a 0.7-metre optical telescope dedicated to candidate active-debris-removal targets and pre-re-entry monitoring, together with a worldwide network of twenty-three all-sky meteor cameras used to characterise re-entry events. Brazil has established a routine operational workflow downloading catalogue data, cross-checking conjunction alerts, and executing collision-avoidance manoeuvres when the calculated probability exceeds a defined threshold supported by its own space-situational-awareness infrastructure and international data-sharing partnerships.

Ukraine similarly participates in multiple Inter-Agency Space Debris Coordination Committee (IADC) working groups, contributing optical observations that support fragmentation studies, photometric characterisation, and re-entry campaigns. Collectively, these measurement advances are steadily narrowing the population of debris that remains genuinely unobservable, even though the medium-sized fragments identified earlier in this review remain a persistent gap.

➤ *Active Debris Removal (ADR) Methods*

• *Drag Augmentation Systems [DAS]:*

Space-environmental approaches aim to alter how debris interacts with its orbital surroundings rather than

relying on direct mechanical capture. Drag augmentation devices such as foam structures, inflatable balloons, or deployable fiber membranes work by significantly enlarging the debris object's surface area relative to its mass. This reduction in ballistic coefficient increases atmospheric drag, hastening orbital decay and eventual re-entry. The method is appealing for its simplicity and lack of active control requirements, though its utility is largely confined to low Earth orbit (LEO), where residual atmospheric density is enough, and deorbit timelines remain sensitive to altitude and solar activity.

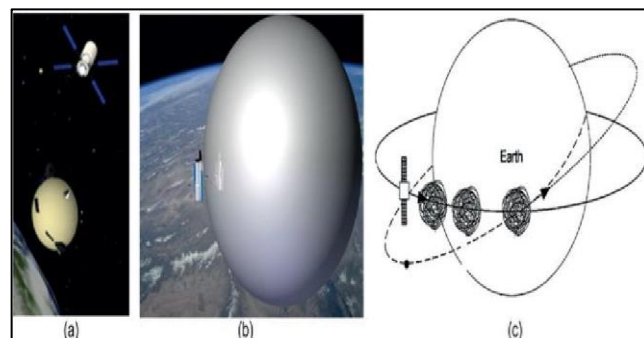


Fig 3 (a) Foam Based, (b) Inflated Based (c) Fiber Based.

• *Adhesive Removal Methods:*

adhesive based removal methods are considered the subset of active debris removal. rather than using nets, harpoons, or tethers, this method depends on sticky or bonding surfaces to capture orbital debris and then maneuver it toward deorbit. Polymer based adhesives use sticky films or gels engineered to withstand vacuum conditions and extreme temperatures. gecko-inspired dry adhesives employ microstructured surfaces that mimic gecko footpads, relying on van der Waals forces to create strong, reusable adhesion without leaving residue.

Electrostatic adhesives generate attractive forces through electrostatic charge, making them particularly effective for smooth metallic debris. Finally, hybrid adhesive pads combine sticky polymers with mechanical micro-hooks, offering enhanced grip on irregular debris surfaces. Together, these approaches highlight the potential of adhesive capture systems for small-to-medium debris, though challenges remain in ensuring reliable adhesion across diverse debris types and orbital conditions.

Table 1 Removal Methods

REMOVAL METHODS	ADVANTAGES	DRAWBACKS	ILLUSTRATIONS
DAS	Companionable with diverse sizes of space debris.	Less efficient.	Foam based Inflated based Fiber based.
CONTACT BASED REMOVAL	Captures even tiny flakes of debris.	Short operational period.	Adhesive removal methods.

III. RESULTS

➤ Debris Inventory and Distribution

Trackable, catalogued debris objects generally those larger than about 10 cm now exceed 50,000, out of roughly 40,000 items actively monitored by surveillance networks, including about 11,000 functioning satellites. Population models estimate that fragments greater than 1 cm number around 1.2 million across all orbital altitudes. The growth pattern has been uneven from the 1960s through the 1990s, debris counts rose in a roughly linear fashion, with slower

accumulation during the early adoption of mitigation guidelines. After 2006, however, growth accelerated sharply due to major fragmentation events, and the upward trend has continued through 2024, when more than 3,000 new trackable objects were added in a single year from breakups alone. Importantly, debris is not uniformly distributed across space, its clusters in heavily used orbital regions, particularly around 550 km in LEO, where debris density now rivals that of active satellites, and in the geostationary belt, both of which represent zones of heightened operational risk.

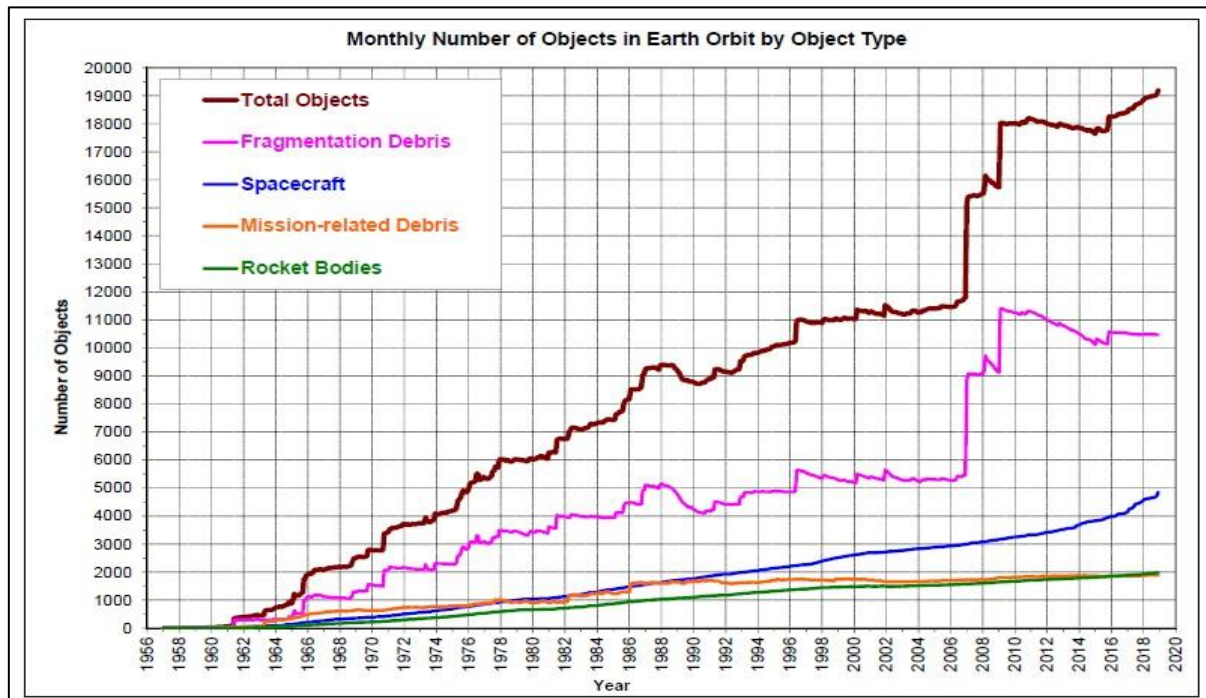


Fig 4 Evaluation of Number of Objects in Space.

➤ Micro Satellite Collision Risk

The study combines monte Carlo simulations with ML methods on representative microsatellite illustrates that semi major axis is the vital factor affecting the collision risk. other parameters like inclination, the mass of the spacecraft, and cross sectional area also play a significant role. depending on the specific orbital setup the operational lifetime of these satellites roughly span around 20 to 100 years. the satellites in LEO have a shorter span than compared to those in GEO and MEO.

➤ Recycling and Reuse methods

• Component Harvesting

Inactive satellites that remain structurally stable can potentially be visited by specialized servicing spacecraft to recover useful components. Parts such as solar panels, batteries, propulsion systems, antennas, and structural sections could be reused in new spacecraft or to repair functioning satellites. Carrying out these operations would require robotic arms, suitable docking mechanisms, and accurate proximity-navigation systems similar to those used in active debris-removal missions.

• Propellant Harvesting

Some satellites reach the end of their missions with a portion of their propellant still remaining because additional fuel is often kept as a safety margin. Instead of allowing this resource to remain unused, future robotic servicing missions could potentially access, collect and reuse it.

• In-Orbit Material Reprocessing

Large pieces of orbital debris, including spent rocket bodies and retired satellite structures, contain materials such as aluminium, titanium, and composites that could potentially be recovered and processed in space. Advanced techniques involving mechanical processing, concentrated solar energy, or other manufacturing methods could convert these materials into usable feedstock. The recovered material could then be used for applications such as 3D-printed components, structural parts, or radiation shielding, reducing the amount of raw material that must be transported from Earth.

• Structural Repurposing

Rather than dismantling every large piece of space debris, some intact structures could be given a new purpose directly in orbit. Retired rocket stages, satellite buses, or other large platforms could potentially be converted into orbital

laboratories, storage facilities, fuel depots, or other useful infrastructure. This approach would make use of existing structures and could reduce the need to launch completely new platforms from Earth.

- *Earth-Return Recycling*

Certain smaller or high-value pieces of space debris could potentially be deliberately brought back to Earth through controlled atmospheric re-entry. Specialized recovery vehicles or capsules could guide these objects toward designated landing or splashdown areas, where their materials could be collected and recycled. Valuable components, including certain metals and electronic materials, could then be recovered through conventional Earth based recycling processes instead of being left as long term orbital debris.

IV. DISCUSSIONS

- No single cleanup technology is the best. What really makes a difference is targeting the objects. Simulations show that removing a few long-lasting satellites or rocket bodies in crowded orbits prevents future chain reactions (Kessler syndrome) far more effectively than randomly removing lots of small fragments.
- Better radars, telescopes, lasers and onboard sensors help us see debris clearly and predict risks. But tracking alone doesn't solve the problem it only shows us where the danger is. With good monitoring the ISS still has to dodge debris several times a year and near-collisions can happen despite being tracked.
- Approaching or grabbing another country's debris could be mistaken as a move and questions of ownership and liability arise. That's why clear agreements, transparency and international cooperation are needed. Efforts like Japan's servicing guidelines and ESA's Zero Debris Charter are steps toward global rules.
- The dangerous debris is the small and medium pieces too many to track and too expensive to remove. Shielding. Doesn't eliminate the risk. Removal missions themselves can create fragments if something goes wrong. And far no mission has fully removed a large derelict satellite; even ESA's ClearSpace-1 is still, in testing. Models show that prevention alone isn't enough anymore—active removal is necessary to stop debris growth in the busiest orbits.

V. SUMMARY AND CONCLUSION

This paper gives a clear in-depth look at space debris tracking where it comes from how it grows and what it does to long-term space travel. It shows that space debris builds up from satellites rocket stages and crashes that break pieces apart making millions of dangerous fragments spread over all kinds of orbits. The study points out the Kessler Syndrome, a situation where collisions make debris than natural decay can take away so taking out debris must be an active job. It says that debris sits mainly in zones such as the low Earth orbit

around 550 kilometres and in the geostationary band, where the danger to working satellites is the highest.

Using Monte Carlo tests and machine-learning tools the research finds that the biggest clues for collision chances are the semi-major axis, the tilt of the orbit the mass and the size of the piece while the shape of the orbit is less important. Different ways to stop and clear debris are examined, such as adding drag, coatings, tethers, nets, lasers and recycling ideas and it stresses how vital global teamwork and laws are. The final message is that no single gadget can fix everything; instead, we must focus on removing the longest-lasting pieces get better tracking and work together worldwide to keep the space environment safe and make space travel last, for many years.

RECOMMENDATIONS

The recommendations from this research emphasize a multi-layered approach to tackling the space debris challenge. Priority should be given to removing large, long-lived objects such as defunct satellites and rocket bodies in congested orbital regions like LEO at 550 km and the GEO belt, since eliminating a few high-risk objects is far more effective than removing numerous small fragments.

At the same time, active debris removal technologies must be advanced, with investment in scalable solutions such as drag augmentation sails, adhesive capture systems, and tether concepts, while continuing demonstration missions like ESA's ClearSpace-1 to achieve full-scale removal. Tracking and sensing capabilities also need to be expanded through radar, optical, and laser-ranging networks to reduce uncertainty, especially for medium-sized debris that remains untracked yet highly hazardous.

Alongside removal, recycling and reuse strategies should be integrated, enabling missions to harvest components, propellant, or materials from defunct satellites and rocket stages for in-orbit manufacturing and infrastructure. On the research front, probabilistic modelling using Monte Carlo and machine-learning methods should be refined to better predict collision risks, focusing on parameters such as semi-major axis, inclination, and cross-sectional area. Efforts must also address small and medium debris through improved shielding, low-cost removal concepts, and in-situ sensors. Finally, prevention and removal must be combined operators should adopt strict end-of-life disposal practices while simultaneously investing in ADR missions, since prevention alone cannot stop debris growth in the most congested orbital bands.

REFERENCES

- [1]. Committee on the Peaceful Uses of Outer Space. (2007, June 15). Space debris mitigation guidelines. United Nations.
- [2]. Wright, D. (2007). Space debris. *Physics Today*, 60(10), 35–40.
- [3]. European Space Agency. (n.d.). Space debris by the numbers. ESA Space Debris Office, ESOC,

- Darmstadt, Germany. *ESA Space Debris by the Numbers*
- [4]. Union of Concerned Scientists. (n.d.). UCS satellite database. *UCS Satellite Database*
 - [5]. Castronuovo, M. M. (2011). Active space debris removal—A preliminary mission analysis and design. *Acta Astronautica*, 69(9–10), 848–859.
 - [6]. Durden, T. (2018, July 9). World's first satellite with harpoon will begin space junk removal test. *ZeroHedge. World's First Satellite With Harpoon Will Begin Space Junk Removal Test*
 - [7]. Viikari, L. (2008). The environmental element in space law: Assessing the present and charting the future. Brill.
 - [8]. Nishida, S., & Kawamoto, S. (2009). Space debris removal system using a small satellite. *Acta Astronautica*, 65(1–2), 95–102.
 - [9]. Garcia, M. (2015, April). Space debris and human spacecraft. National Aeronautics and Space Administration. *NASA Space Debris and Human Spacecraft*
 - [10]. National Aeronautics and Space Administration. (2019, May). Orbital debris quarterly news, 23(1–2).
 - [11]. Liou, J.-C., & Johnson, N. L. (2008). Instability of the present LEO satellite populations. *Advances in Space Research*, 41(7), 1047–1056.
 - [12]. O'Connor, B. (2008). Handbook for limiting orbital debris (NASA Handbook 8719.14). National Aeronautics and Space Administration.
 - [13]. Kessler, D. J., Johnson, N. L., Liou, J.-C., & Matney, M. (2010). The Kessler syndrome: Implications to future space operations. *Advances in the Astronautical Sciences*, 137, 47–54.
 - [14]. McKnight, D. (2010). Pay me now or pay me more later: Start the development of active orbital debris removal now. *Advanced Maui Optical and Space Surveillance Technologies Conference*.
 - [15]. Tonry, J., & Onaka, P. (2009). *Advanced Maui Optical and Space Surveillance Technologies Conference proceedings*. Advanced Maui Optical and Space Surveillance Technologies Conference.
 - [16]. Klinkrad, H. (2002). Monitoring space: Efforts made by European countries. *International Colloquium on Europe and Space Debris*, France.
 - [17]. Colmenarejo, P., Binet, G., Strippoli, L., Peters, T. V., & Graziano, M. (2013). GNC aspects for active debris removal. *Proceedings of the 2nd CEAS Specialist Conference on Guidance, Navigation and Control (EuroGNC 2013)*, The Netherlands.
 - [18]. Nishida, S.-I., & Kikuchi, N. (2014). A scenario and technologies for space debris removal. *12th International Symposium on Artificial Intelligence, Robotics and Automation in Space*.
 - [19]. Chu, Z., Di, J., & Cui, J. (2017). Analysis of the effect of attachment point bias during large space debris removal using a tethered space tug. *Acta Astronautica*, 139, 34–41.
 - [20]. Shuvalov, V. A., Gorev, N. B., Tokmak, N. A., & Kochubei, G. S. (2017). Physical simulation of the long-term dynamic action of a plasma beam on a space debris object. *Acta Astronautica*, 132, 97–102.
 - [21]. Wen, Q., et al. (2017). Removing small scale space debris by using a hybrid ground and space based laser system. *Optik*, 141, 105–113.
 - [22]. Zhang, F., et al. (2017). Dynamics modeling and model selection of space debris removal via the tethered space robot. *Proceedings of the Institution of Mechanical Engineers, Part G: Journal of Aerospace Engineering*, 231(10), 1873–1897.
 - [23]. Takahashi, K., et al. (2018). Demonstrating a new technology for space debris removal using a bi-directional plasma thruster. *Scientific Reports*, 8, Article 14417.
 - [24]. Benvenuto, R., Lavagna, M., Lunghi, P., et al. (2017). Tethered-tugs for active debris removal: Microgravity experimental validation of dynamics and control. *Proceedings of the 7th European Conference on Space Debris*, Darmstadt, Germany.
 - [25]. Liedahl, D. A., et al. (2013). Pulsed laser interactions with space debris: Target shape effects. *Advances in Space Research*, 52(5), 895–915.
 - [26]. Andrenucci, M., Pergola, P., & Ruggiero, A. (2011). Active removal of space debris—Expanding foam application for active debris removal. *European Space Agency Final Report*.
 - [27]. Nock, K., et al. (2010). Gossamer orbit lowering device (GOLD) for safe and efficient de-orbit. *AIAA/AAS Astrodynamics Specialist Conference*.
 - [28]. Wright, R. J. (2013). Orbital debris mitigation system and method (U.S. Patent No. 8,567,725). U.S. Patent and Trademark Office.
 - [29]. Missel, W. J. (2013). Active space debris removal using capture and ejection [Doctoral dissertation].
 - [30]. Choi, H. S., & Pappa, R. S. (2012). Assessment study of small space debris removal by laser satellites. *Recent Patents on Space Technology*, 2(2), 116–122.
 - [31]. Ishige, Y., Kawamoto, S., & Kibe, S. (2004). Study on electrodynamic tether system for space debris removal. *Acta Astronautica*, 55(11), 917–929.
 - [32]. Samanta Roy, R. I., Hastings, D. E., & Ahedo, E. (1992). System analysis of electrodynamic tethers. *Journal of Spacecraft and Rockets*, 29(3), 415–423.
 - [33]. Ohkawa, Y., et al. (2015). Preparation for on-orbit demonstration of electrodynamic tether on HTV. *Proceedings of the Joint Conference of the 30th International Symposium on Space Technology and Science, 34th International Electric Propulsion Conference, and 6th Nano-Satellite Symposium*, 4–10.
 - [34]. Wormnes, K., et al. (2013). ESA technologies for space debris remediation. *Proceedings of the 6th European Conference on Space Debris*. ESA Communications.
 - [35]. Williams, P. (2005). Optimal orbit transfer with electrodynamic tether. *Journal of Guidance, Control, and Dynamics*, 28(3), 369–372.
 - [36]. Missel, J., & Mortari, D. (2013). Path optimization for Space Sweeper with Sling-Sat: A method of active space debris removal. *Advances in Space Research*, 52(7), 1339–1348.
 - [37]. Gregory, D., Mergen, J., & Ridley, A. (2011). Space debris elimination (SpaDE) Phase I final report. National Aeronautics and Space Administration.

- [38]. Stuart, J., Howell, K., & Wilson, R. (2016). Application of multi-agent coordination methods to the design of space debris mitigation tours. *Advances in Space Research*, 57(8), 1680–1690.
- [39]. Okada, N. (2014). Active debris removal using carrier and multiple deorbiting kits. 3rd European Workshop on Space Debris Modelling and Remediation.
- [40]. Missel, J., & Mortari, D. (2013). Removing space debris through sequential captures and ejections. *Journal of Guidance, Control, and Dynamics*, 36(3), 743–752.
- [41]. Zinner, N., Williamson, A., Brenner, K., Curran, J. B., Isaak, A., Knoch, M., et al. (2011). Junk hunter: Autonomous rendezvous, capture, and de-orbit of orbital debris. AIAA SPACE 2011 Conference & Exposition, Long Beach, CA, United States.
- [42]. Gregory, D., & Mergen, J. (2014). Space debris removal using upper atmosphere and vortex generator (U.S. Patent No. 8,657,235). U.S. Patent and Trademark Office.
- [43]. Lappas, V., Adeli, N., Visagie, L., Fernandez, J., Theodorou, T., Steyn, W., & Perren, M. (2011). CubeSail: A low cost CubeSat based solar sail demonstration mission. *Advances in Space Research*, 48(11), 1890–1901.
- [44]. Bombardelli, C., & Peláez, J. (2011). Ion beam shepherd for contactless space debris removal. *Journal of Guidance, Control, and Dynamics*, 34(3), 916–920.
- [45]. Retat, B., Bischof, et al. (2012). Net capture system: A potential orbital space debris removal system. 2nd European Workshop on Active Debris Removal, CNES Headquarters, Paris, France.
- [46]. Ionin, A. A., Kudryashov, S. I., & Seleznev, L. V. (2010). Near-critical phase explosion promoting breakdown plasma ignition during laser ablation of graphite. *Physical Review E*, 82(1), 016404.
- [47]. Merino, M., Ahedo, E., Bombardelli, C., Urrutxua, H., & Peláez, J. (2013). Ion beam shepherd satellite for space debris removal. *Progress in Propulsion Physics*, 4, 789–802.
- [48]. DeLuca, L. T., Bernelli, F., et al. (2013). Active space debris removal by a hybrid propulsion module. *Acta Astronautica*, 91, 20–33.
- [49]. Phipps, C. (2014). A laser-optical system to re-enter or lower low Earth orbit space debris. *Acta Astronautica*, 93, 418–429.
- [50]. Sahara, H. (2014). Evaluation of a satellite constellation for active debris removal. *Acta Astronautica*, 105, 136–144. <https://doi.org/10.1016/j.actaastro.2014.08.026>
- [51]. Aslanov, V., & Yuditsev, V. (2018). Motion control of space tug during debris removal by a Coulomb force. *Journal of Guidance, Control, and Dynamics*, 41(7), 1476–1484.
- [52]. Singh, P., Chand, D. S., Pal, S., & Mishra, A. (2020). Study of current scenario & removal methods of space debris. *International Journal of Mechanical and Production Engineering Research and Development*, 10(Special Issue), 223–236.
- [53]. European Space Agency. (2025). ESA space environment report 2025. ESA Space Safety. *ESA Space Environment Report*
- [54]. European Space Agency. (n.d.). ESA advances ClearSpace-1 development. ESA Space Safety. *ESA ClearSpace-1*
- [55]. ClearSpace. (2026, January 12). ESA and ClearSpace start PRELUDE, a new chapter. *ClearSpace*
- [56]. Medhin, Y., & Servadio, S. (2025). The sustainability of the LEO orbit capacity via risk-driven active debris removal (arXiv:2507.16101). arXiv. <https://doi.org/10.48550/arXiv.2507.16101>.